

**APPROVED FOR PUBLIC RELEASE  
DISTRIBUTION UNLIMITED**

**AFPTEF PROJECT NO. 04-P-111**

**SUSAN J. EVANS**

**Mechanical Engineer**

**DSN 787-7445  
Comm (937) 257-7445**

**Qualification Testing of the B-52 Nose Radome  
Container, CNU-680/E**

**HQ AFMC/LSO/LOP  
AIR FORCE PACKAGING TECHNOLOGY & ENGINEERING FACILITY  
WRIGHT PATTERSON AFB, OH 45433-5540  
January 2006**

**NOTICE**

When government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related government procurement operation, the United States Government thereby incurs no responsibility whatsoever; and the fact that the government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto. This report is not to be used in whole or in part for advertising or sales purposes.

**AFPTEF PROJECT NO.: 04-P-111**  
**TITLE: B-52 Nose Radome Container**

**ABSTRACT**

The objective of this test series was to qualify the B-52 Nose Radome Shipping and Storage container, AFPTEF project number 04-P-111, for production release by AFMC LSO/LOP. The container is a sealed, reusable, aluminum container engineered for the physical and environmental protection of the B-52 Nose Radome during worldwide transportation and storage.

The test plan referenced SAE ARP 1967 and ASTM D 4169. All tests were performed at the Air Force Packaging Technology & Engineering Facility (AFPTEF), AFMC LSO/LOP, 5215 Thurlow St, Bldg 70, Wright-Patterson AFB OH 45433-5540.

**Total Project Hours: 65**

**TESTED BY:**

SUSAN J. EVANS  
Mechanical Engineer

Susan J. Evans

**REVIEWED BY:**

JOEL A. SULLIVAN  
Mechanical Engineer

Joel A. Sullivan

**APPROVED BY:**

ROBBIN L. MILLER  
Chief, AFPTEF

Robbin L. Miller

## TABLE OF CONTENTS

	<u>PAGE</u>
Abstract	9
Table of Contents	10
Introduction	11
Background	11
Requirements	11
Development	11
Design	11
Testing	12
Test Specimen	12
Test Load	12
Test Procedures	12
Instrumentation and Equipment	12
Test Sequences	13
Test Results	15
Project Conclusions	15
Test Data	16

## **INTRODUCTION**

### **BACKGROUND**

The B-52 Logistics Management office located at Tinker AFB requested that the Air Force Packaging Technology and Engineering Facility (AFPTEF) develop a long-life aluminum container for the B-52 Nose Radome. This container is a replacement for the current packaging system consisting of a wood box which degrades readily in outdoor long-term storage and generally provides inadequate protection for the radome.

### **REQUIREMENTS**

The container test plan (see Appendix 1) was developed for qualifying the container for worldwide transportation and storage.

## **DEVELOPMENT**

### **DESIGN OF THE CONTAINER**

The B-52 Nose Radome Shipping and Storage Container is a sealed, reusable, aluminum container engineered for the physical and environmental protection of the B-52 Nose Radome during worldwide transportation and storage. The container consists of a base and completely removable cover equipped with the special features listed below. A silicone rubber gasket and cam-over-center latches create a watertight seal at the base/lid interface.

An enclosed four-way forklift access aluminum base is welded to the container bottom. An aluminum cradle system is mounted on helical steel isolators, which in turn are mounted to the interior container sides. The isolators limit the transmission of shock to the radome to 50 Gs. A lifting frame attaches to the radome using four pins and two hook locks in the aft, and by two pins in the forward. The lifting frame and radome are then attached to the cradle system using four clamps. Container external dimensions are 135.2 inches length, 104.1 inches width, and 105.1 inches height. Container empty weight is 2402 pounds, and 2620 pounds with the radome in place.

RADOME CONTAINER FEATURES	
PRESSURE RELIEF VALVE	FIVE
HUMIDITY INDICATOR	ONE
DESICCANT PORT	TWO
DOCUMENT RECEPTACLE	NONE
FORKLIFT TABLE	YES
COVER LATCHES	28
COVER LIFT HANDLES	NONE
COVER LIFT RINGS	ONE
COVER TETHER RINGS	FOUR
BASE LIFT HANDLES	NONE
BASE TIREDOWN RINGS	SIX
STACKING INTERFACE	NO

## **TESTING**

### **TEST SPECIMEN**

The test specimen was an aluminum container manufactured by AFPTEF.

### **TEST LOAD**

The test load was an unserviceable, reparable, B-52 Nose Radome.

### **TEST PROCEDURES**

The radome container was tested in accordance with the Air Force Packaging Technology & Engineering Facility (AFPTEF) modified long life container test plan (Appendix 1).

The test plan primary references were ASTM D 4169 and SAE ARP 1967. The test methods specified in this test plan constituted the procedure for performing the tests on the radome container. The performance criteria for evaluation of container acceptability were specified at 50 Gs maximum and an initial and final leak rate of 0.05 psi/hr at 1.0 psi. These tests are commonly applied to special shipping containers providing rough handling protection to sensitive items. The tests were performed at AFPTEF, AFMC LSO/LOP, 5215 Thurlow St, Wright-Patterson AFB, OH 45433-5540.

## **INSTRUMENTATION AND EQUIPMENT**

### **CARRIER FACE IDENTIFICATION**

The correlation between container sides and container features for test purposes was as follows (See Appendix 3, Figure 5):

<b>DESIGNATED SIDE</b>	<b>CONTAINER FEATURE</b>	<b>NUMBER</b>
Top	Cover Top	1
Aft	Desiccant Port	4
Forward	Opposite Aft	2
Left	Left (Long) Side	6
Right	Right (Long) Side	5
Bottom	Base Bottom	3

### **ITEM INSTRUMENTATION**

The test load was instrumented with a piezoelectric triaxial accelerometer mounted as close as possible to the radome's center of mass. Accelerometer positive axis orientations were as follows:

X Axis - Directed through container Top and Bottom (Vertical motion).

Y Axis - Directed through container Forward and Aft sides (Longitudinal motion).

Z Axis - Directed through container Left and Right sides (Transverse motion).

PRESSURE TEST EQUIPMENT - Test sequences 1 & 5

EQUIPMENT	MANUFACTURER	MODEL	SN	CAL. DATE
Digital Manometer	Yokogawa	2655	82DJ6009	June 05

ROUGH HANDLING TEST EQUIPMENT - Test sequences 2 & 4.

EQUIPMENT	MANUFACTURER	MODEL	SN	CAL. DATE
Shock Amplifier	Endevco	2775A	ER34	NA
Shock Amplifier	Endevco	2775A	ER33	NA
Shock Amplifier	Endevco	2775A	EL81	NA
Radome Accelerometer	Endevco	2223D	FL46	Sep 04
Radome Accelerometer	Endevco	2228C	16471	Dec 05
Data Acquisition	GHI Systems	CAT	Ver. 2.7.1	N/A

VIBRATION TEST EQUIPMENT - Test sequence 3.

EQUIPMENT	MANUFACTURER	MODEL	SN	CAL. DATE
Transportation Data Recorder	IST	EDR-3	9009280082	NA

For the vibration test only, the test load was instrumented with a second piezoelectric triaxial accelerometer, also mounted as close as possible to the radome's center of mass. Accelerometer positive axis orientations were as follows:

X Axis - Directed through container Forward and Aft sides (Longitudinal motion).

Y Axis - Directed through container Top and Bottom (Vertical motion).

Z Axis - Directed through container Left and Right sides (Transverse motion).

## TEST SEQUENCES

Note: All test sequences were performed at ambient temperature and humidity.

### TEST SEQUENCE 1 - SAE ARP 1967, para. 4.5.2 – Containers, Shipping & Storage, Reusable, Leak Test

The left desiccant port cover was removed and replaced with a port cover modified for attachment of the digital manometer and vacuum/pressure pump lines. The right desiccant port cover was replaced with a cover modified for attachment of a high-pressure air line. The container was closed and sealed. The leak test was conducted in accordance with the above specification, at ambient temperature and pressure. The pneumatic pressure leak technique was used to pressurize the container to a minimum test pressure of 1.0 psi (See Appendix 3, Figure 8).

**TEST SEQUENCE 2 - SAE ARP 1967, para. 4.5.3 – Containers, Shipping & Storage, Reusable, ASTM D4169, Schedule A, para. 10.3.3.1(3), Assurance Level I – Rotational Drops (ASTM D6179, Methods A & B)**

An Assurance Level I drop height of 12 in. was used to perform four corner and four edge drops onto a one-inch thick steel plate (See Appendix 3, Figures 9 and 10).

**TEST SEQUENCE 3 - Over-the-Road Vehicle Vibration Test**

Due to the container size, the standard vibration tests (SAE ARP 1967, para. 4.5., Vibration Test; and ASTM D4169, Schedule E, para. 12.5, Vehicle Vibration, Sine Test Option, Resonance Dwell) could not be performed on this prototype. Instead, an over-the-road vehicle vibration test, intended to duplicate real-world conditions as closely as possible, was performed. The container was placed on the wood deck of a tractor trailer, and held in place using cargo straps attached to the tie-down rings. (See Appendix 3, Figure 12).

The vibration and acceleration experienced by the radome were recorded for a period of 1.5 hours using an EDR-3 transportation environment data recorder. The container was transported over a variety of surfaces including: gravel; abandoned and broken asphalt paving (10 mph to 30 mph); concrete and asphalt interstate highway (55 mph), both newly-paved and several years old; 2-lane and 4-lane asphalt state highways (35 mph to 55 mph); and various 2-lane asphalt roads on Wright-Patterson AFB at speeds ranging from 10 mph to 45 mph (See Appendix 3, Figure 12).

**TEST SEQUENCE 4 - SAE ARP 1967, para. 4.5.6 – Containers, Shipping & Storage, Reusable ASTM D4169, Schedule A, para. 10.3.3.1(4), Assurance Level I – Lateral Impacts (ASTM D880, Procedure A)**

Upon completion of test sequence 3, the container was transported to Tinker AFB for a fit and function check. Upon its return, the loaded container was placed on the test apparatus and impacted. The container impact velocity was 2.13 m/sec. Only the aft and forward container sides could be impacted (one time each) due to the size limitations of the test apparatus (See Appendix 3, Figure 11).

**TEST SEQUENCE 5 - SAE ARP 1967, para. 4.5.2 – Containers, Shipping & Storage, Reusable, Leak Test**

The left desiccant port cover was removed and replaced with a port cover modified for attachment of the digital manometer and vacuum/pressure pump lines. The right desiccant port cover was replaced with a cover modified for attachment of a high-pressure air line. The container was closed and sealed. The leak test was conducted in accordance with the above specification, at ambient temperature and pressure. The pneumatic pressure leak technique was used to pressurize the container to a minimum test pressure of 1.0 psi. (See Appendix 3, Figure 8)

## **TEST RESULTS**

### **Test Sequence 1 – Leak Test**

The container passed the leak test with a leak rate less than the maximum allowed rate of 0.05 psi per hour.

### **Test Sequence 2 – Rough Handling: Rotational Drops**

There was no noticeable damage to either the container or item. There was no flattening of clay placed on the radome at points where excessive swaying of the support frame might have allowed the radome to impact the container lid. The maximum recorded impacts, after filtering at 200 Hz to reduce excessive ringing (buzzing) from the cradle frame and radome, ranged from 18 Gs to 27 Gs, well below the item fragility of 50 Gs. Without filtering the G-levels remained below 35 Gs (See Test Data, Table 1 and Graphs 1 - 8). The container met the test requirements.

**Test Sequence 3 – Over-the-Road Vibration Test.** No accelerations greater than 5 Gs were recorded for any axis. The vibration recordings for the events with the highest G levels, as well as a sampling of other events, were observed; however there were no signs of increasing vibration amplitude or of anything else that would cause concern. The container met the test requirements (See Test Data, Table 2).

### **Test Sequence 4 – Rough Handling: Lateral Impacts**

No noticeable damage occurred to the container or item. The item did not make contact with any interior container surfaces during testing. The maximum recorded impacts, after filtering at 200 Hz to compensate for excessive ringing, ranged from 11 Gs to approximately 20 Gs, all below the item fragility of 50 Gs. Without filtering the maximum G-levels ranged from 14 Gs to 30 Gs (See Test Data, Table 1 and Graph 9). Although the recorded data file for the aft side impact was not recorded, it is known to have been below 30 Gs without filtering. The container met the test requirements.

### **Test Sequence 5 – Leak Test**

The container passed the leak test with a leak rate less than the maximum allowed rate of 0.05 psi.

## **PROJECT CONCLUSIONS**

No damage occurred during the above testing to the container, mounting system or test item. There was no evidence of any contact from impacts between the radome and the container walls or lid. All impact levels are well below the item fragility limit of 50 Gs. Therefore, the container and mounting system do provide adequate protection for the radome.

**TABLE 1. Impact Test Summary**

IMPACT TYPE	TEST TEMPERATURE	IMPACT LOCATION	RESULTANT PEAK G
ROTATIONAL - CORNER	ambient	forward-left	21
ROTATIONAL - CORNER	ambient	forward-right	27
ROTATIONAL - CORNER	ambient	aft-left	18
ROTATIONAL - CORNER	ambient	aft-right	24
ROTATIONAL - EDGE	ambient	forward-bottom	23
ROTATIONAL - EDGE	ambient	aft-bottom	19
ROTATIONAL - EDGE	ambient	left-bottom	20
ROTATIONAL - EDGE	ambient	right-bottom	25
LATERAL IMPACT - FACE	ambient	forward	17
LATERAL IMPACT - FACE	ambient	aft*	*
LATERAL IMPACT - FACE	ambient	left	NA
LATERAL IMPACT - FACE	ambient	right	NA

\* Test data not recorded; however, the resultant peak G for this impact is known to have been less than 30 Gs without filtering.

**TABLE 2. Tabulated Vibration (Impact) Test Data.**

Tue Aug 23 12:54:23 2005

**Tabulated Impact Report**

page 0001

File: B52 RADM  
 Sorted by: Time  
 Total events: 200  
 Table of: Acceleration waveforms  
 Report Subject: B5 Radome  
 Over-the-Road Vibration/Impact Test.

EV #	time	Max X g	Max Y g	Max Z g
1 00	08/23/105 09:24:25	1.231	1.057	0.880
2 00	08/23/105 09:24:33	2.556	2.017	1.466
3 00	08/23/105 09:24:42	1.420	1.441	1.075
4 00	08/23/105 09:24:51	0.852	1.057	0.684
5 00	08/23/105 09:24:59	1.041	1.057	0.489
6 00	08/23/105 09:25:08	1.041	1.057	0.684
7 00	08/23/105 09:25:16	1.231	1.249	0.880
8 00	08/23/105 09:25:24	1.609	1.633	0.880
9 00	08/23/105 09:26:16	1.420	1.441	1.857
10 00	08/23/105 09:26:24	1.420	2.017	2.053
11 00	08/23/105 09:32:05	1.041	1.825	1.075
12 00	08/23/105 09:32:14	1.231	1.057	0.880
13 00	08/23/105 09:32:23	1.041	1.441	1.662
14 00	08/23/105 09:32:31	1.041	1.057	0.684
15 00	08/23/105 09:32:40	1.420	1.441	2.053
16 00	08/23/105 09:33:14	1.041	1.249	0.684
17 00	08/23/105 09:33:31	2.367	2.401	3.030
18 00	08/23/105 09:33:40	1.231	1.633	0.880
19 00	08/23/105 09:33:49	1.231	1.825	1.075
20 00	08/23/105 09:33:57	1.420	1.633	2.835
21 00	08/23/105 09:34:06	1.799	1.633	2.248
22 00	08/23/105 09:34:15	1.231	1.441	1.662
23 00	08/23/105 09:34:23	2.177	2.209	1.857
24 00	08/23/105 09:34:49	1.420	1.441	1.271
25 00	08/23/105 09:34:57	1.231	1.249	0.880
26 00	08/23/105 09:35:06	1.231	1.825	1.662
27 00	08/23/105 09:35:15	1.420	1.825	2.248
28 00	08/23/105 09:35:32	1.231	1.441	2.053
29 00	08/23/105 09:35:40	2.177	2.209	2.053
30 00	08/23/105 09:35:49	1.420	1.633	2.248
31 00	08/23/105 09:35:57	1.420	1.825	1.466
32 00	08/23/105 09:36:06	1.420	1.441	1.466
33 00	08/23/105 09:36:14	2.745	2.017	1.466
34 00	08/23/105 09:36:23	1.799	1.633	1.662
35 00	08/23/105 09:36:31	5.775	4.706	3.421
36 00	08/23/105 09:36:40	3.124	2.593	3.030
37 00	08/23/105 09:36:48	1.609	2.017	1.662
38 00	08/23/105 09:36:57	1.420	1.825	2.053
39 00	08/23/105 09:37:23	1.231	1.441	1.662
40 00	08/23/105 09:37:31	2.177	1.825	2.248
41 00	08/23/105 09:37:40	1.799	2.209	2.444
42 00	08/23/105 09:37:48	2.367	2.209	2.444
43 00	08/23/105 09:37:57	1.799	1.825	4.203
44 00	08/23/105 09:38:05	2.177	1.825	2.053
45 00	08/23/105 09:38:14	1.420	1.633	1.857
46 00	08/23/105 09:38:57	1.231	1.249	0.880

X: for A  
 Y: vert  
 Z: trans

**TABLE 2. Tabulated Vibration (Impact) Test Data (Continued).**

Tue Aug 23 12:54:23 2005

**Tabulated Impact Report**

page 0002

EV #	time	Max X g	Max Y g	Max Z g
47	00 08/23/105 09:39:14	1.231	1.441	2.248
48	00 08/23/105 09:39:23	1.231	1.441	1.271
49	00 08/23/105 09:40:05	1.799	2.017	2.444
50	00 08/23/105 09:40:14	1.609	1.441	1.662
51	00 08/23/105 09:42:47	1.799	1.057	1.662
52	00 08/23/105 09:43:04	1.231	1.057	1.075
53	00 08/23/105 09:43:13	1.231	1.249	1.075
54	00 08/23/105 09:43:30	1.041	2.017	1.271
55	00 08/23/105 09:43:38	1.420	1.441	1.662
56	00 08/23/105 09:43:47	1.799	2.593	2.639
57	00 08/23/105 09:43:55	1.041	1.057	0.880
58	00 08/23/105 09:44:04	2.556	2.593	2.444
59	00 08/23/105 09:45:46	1.420	1.441	1.271
60	00 08/23/105 09:45:55	1.609	2.209	1.466
61	00 08/23/105 09:46:20	1.799	1.441	1.662
62	00 08/23/105 09:46:37	1.420	1.057	0.880
63	00 08/23/105 09:46:46	1.420	1.057	0.880
64	00 08/23/105 09:46:54	2.556	2.401	2.053
65	00 08/23/105 09:47:11	1.988	1.441	0.880
66	00 08/23/105 09:47:20	1.420	1.441	1.662
67	00 08/23/105 09:48:36	1.041	0.864	1.466
68	00 08/23/105 09:50:02	1.609	2.017	1.662
69	00 08/23/105 09:50:19	1.420	1.249	1.662
70	00 08/23/105 09:50:27	1.231	0.672	1.466
71	00 08/23/105 09:51:02	1.799	1.441	1.857
72	00 08/23/105 09:52:10	1.420	1.249	1.466
73	00 08/23/105 09:55:52	2.177	1.825	1.857
74	00 08/23/105 09:56:00	1.231	1.057	0.880
75	00 08/23/105 09:56:17	1.231	1.057	1.075
76	00 08/23/105 09:56:25	2.935	2.593	2.639
77	00 08/23/105 09:56:51	1.231	1.249	1.075
78	00 08/23/105 09:57:00	1.420	2.209	2.248
79	00 08/23/105 09:57:08	1.420	1.441	2.053
80	00 08/23/105 09:57:42	1.041	0.864	0.880
81	00 08/23/105 09:57:51	2.177	2.209	3.030
82	00 08/23/105 09:58:51	1.231	0.864	0.684
83	00 08/23/105 09:59:08	1.231	1.633	1.271
84	00 08/23/105 09:59:33	1.231	0.864	1.466
85	00 08/23/105 09:59:42	0.852	0.672	0.880
86	00 08/23/105 09:59:51	1.988	2.209	3.030
87	00 08/23/105 09:59:59	1.420	2.017	2.053
88	00 08/23/105 10:01:16	1.420	1.441	1.075
89	00 08/23/105 10:01:42	1.420	1.441	2.248
90	00 08/23/105 10:01:50	1.799	1.825	1.857
91	00 08/23/105 10:02:24	1.041	1.057	0.684
92	00 08/23/105 10:03:49	1.231	1.057	1.466
93	00 08/23/105 10:03:58	1.420	1.441	1.466
94	00 08/23/105 10:06:48	1.799	1.633	1.271
95	00 08/23/105 10:08:30	1.799	2.017	2.444
96	00 08/23/105 10:13:28	1.041	0.864	0.684
97	00 08/23/105 10:13:36	1.231	0.672	0.684
98	00 08/23/105 10:13:45	1.231	0.672	0.880
99	00 08/23/105 10:14:02	1.231	1.249	0.880
100	00 08/23/105 10:15:01	1.231	1.249	0.880

**TABLE 2. Tabulated Vibration (Impact) Test Data (Continued).**

Tue Aug 23 12:54:23 2005      Tabulated Impact Report      page 0003

EV #	time	Max X g	Max Y g	Max Z g
101	00 08/23/105 10:15:10	1.041	1.057	0.880
102	00 08/23/105 10:17:01	1.609	2.017	2.444
103	00 08/23/105 10:17:52	1.041	0.864	0.880
104	00 08/23/105 10:18:09	1.609	1.441	2.053
105	00 08/23/105 10:22:58	1.231	1.249	0.880
106	00 08/23/105 10:23:49	1.609	1.825	2.248
107	00 08/23/105 10:23:57	1.041	1.057	1.271
108	00 08/23/105 10:24:14	1.041	1.249	1.466
109	00 08/23/105 10:24:23	1.231	1.057	0.880
110	00 08/23/105 10:24:31	1.988	1.825	2.248
111	00 08/23/105 10:24:40	1.231	1.441	1.075
112	00 08/23/105 10:25:05	1.041	1.057	1.271
113	00 08/23/105 10:25:23	1.041	1.249	1.466
114	00 08/23/105 10:25:40	1.231	0.864	1.662
115	00 08/23/105 10:25:57	1.041	0.864	1.075
116	00 08/23/105 10:26:14	1.799	1.633	1.662
117	00 08/23/105 10:26:22	1.231	1.825	2.053
118	00 08/23/105 10:26:56	1.231	0.864	0.880
119	00 08/23/105 10:28:04	1.231	1.441	1.271
120	00 08/23/105 10:28:30	1.231	1.249	1.075
121	00 08/23/105 10:29:21	1.420	1.249	0.880
122	00 08/23/105 10:29:46	1.420	0.864	1.662
123	00 08/23/105 10:31:37	2.367	2.401	2.639
124	00 08/23/105 10:31:45	1.420	1.441	1.075
125	00 08/23/105 10:31:54	1.231	0.864	0.880
126	00 08/23/105 10:32:11	1.609	1.441	1.271
127	00 08/23/105 10:32:19	1.609	2.017	2.053
128	00 08/23/105 10:32:45	1.420	1.441	1.271
129	00 08/23/105 10:32:53	1.231	1.249	0.880
130	00 08/23/105 10:33:02	1.041	1.249	1.271
131	00 08/23/105 10:33:10	2.177	2.401	2.639
132	00 08/23/105 10:33:27	2.745	2.785	3.421
133	00 08/23/105 10:34:10	1.041	1.249	0.684
134	00 08/23/105 10:36:34	1.799	1.441	1.271
135	00 08/23/105 10:36:43	1.609	2.017	1.271
136	00 08/23/105 10:36:51	1.799	1.441	1.271
137	00 08/23/105 10:37:00	1.799	2.017	1.662
138	00 08/23/105 10:37:08	2.556	1.633	1.466
139	00 08/23/105 10:37:17	1.799	1.249	1.662
140	00 08/23/105 10:38:25	1.231	0.864	0.880
141	00 08/23/105 10:38:33	2.367	2.209	2.248
142	00 08/23/105 10:40:15	1.420	1.825	1.662
143	00 08/23/105 10:40:49	1.799	1.633	2.348
144	00 08/23/105 10:40:58	1.041	1.441	1.662
145	00 08/23/105 10:41:15	1.041	0.864	0.880
146	00 08/23/105 10:41:23	1.041	0.864	1.271
147	00 08/23/105 10:41:32	1.041	0.864	1.075
148	00 08/23/105 10:41:58	1.420	1.057	1.662
149	00 08/23/105 10:42:06	1.231	1.825	2.053
150	00 08/23/105 10:42:32	1.231	1.057	0.684
151	00 08/23/105 10:42:40	1.231	0.864	1.466
152	00 08/23/105 10:44:13	2.556	2.785	3.421
153	00 08/23/105 10:45:13	1.420	1.825	2.053
154	00 08/23/105 10:50:54	1.420	1.633	1.857

**TABLE 2. Tabulated Vibration (Impact) Test Data (Continued).**

			1.023	1.031	
			1.607	1.611	
Tue Aug 23 12:54:23 2005		Tabulated Impact Report			page 0004
EV #	time	Max X g	Max Y g	Max Z g	
155	00 08/23/105 10:51:03	1.231	1.249	0.880	
156	00 08/23/105 10:51:11	1.231	1.249	1.075	
157	00 08/23/105 10:51:28	1.231	1.249	1.075	
158	00 08/23/105 10:51:37	1.041	1.249	0.880	
159	00 08/23/105 10:51:45	1.799	2.209	2.835	
160	00 08/23/105 10:52:36	1.420	1.633	3.030	
161	00 08/23/105 10:52:53	1.041	1.057	1.075	
162	00 08/23/105 10:53:36	1.231	1.057	1.075	
163	00 08/23/105 10:53:44	1.231	1.057	0.880	
164	00 08/23/105 10:54:27	1.041	1.441	1.271	
165	00 08/23/105 10:55:26	1.420	1.633	1.075	
166	00 08/23/105 10:55:35	1.799	1.825	1.662	
167	00 08/23/105 10:55:43	1.609	1.825	1.466	
168	00 08/23/105 10:56:09	1.988	1.825	1.662	
169	00 08/23/105 10:56:51	1.041	1.057	0.880	
170	00 08/23/105 10:57:51	1.420	1.249	1.466	
171	00 08/23/105 10:58:59	1.041	0.864	0.880	
172	00 08/23/105 10:59:07	1.231	1.057	1.075	
173	00 08/23/105 10:59:16	1.041	1.633	1.662	
174	00 08/23/105 10:59:24	1.799	1.633	1.466	
175	00 08/23/105 10:59:50	1.041	1.057	1.075	
176	00 08/23/105 10:59:58	1.041	0.864	0.684	
177	00 08/23/105 11:00:07	1.041	1.057	0.880	
178	00 08/23/105 11:00:16	1.041	1.057	1.857	
179	00 08/23/105 11:00:24	1.231	1.441	1.662	
180	00 08/23/105 11:00:33	1.799	0.864	0.880	
181	00 08/23/105 11:00:41	1.041	0.864	0.880	
182	00 08/23/105 11:01:15	1.041	1.057	0.684	
183	00 08/23/105 11:01:24	1.231	1.249	1.271	
184	00 08/23/105 11:01:32	1.799	1.249	2.053	
185	00 08/23/105 11:01:49	2.177	2.017	2.639	
186	00 08/23/105 11:04:22	1.231	1.057	0.880	
187	00 08/23/105 11:04:31	1.231	1.249	1.075	
188	00 08/23/105 11:04:39	1.988	2.401	1.857	
189	00 08/23/105 11:04:48	2.177	2.401	1.857	
190	00 08/23/105 11:04:56	1.231	1.249	1.271	
191	00 08/23/105 11:05:05	2.556	2.785	2.835	
192	00 08/23/105 11:05:13	1.231	1.249	1.466	
193	00 08/23/105 11:05:22	1.420	1.441	1.271	
194	00 08/23/105 11:07:04	1.231	0.864	1.662	
195	00 08/23/105 11:08:54	1.420	1.441	1.857	
196	00 08/23/105 11:09:37	1.420	1.249	0.880	
197	00 08/23/105 11:09:45	1.420	1.633	1.857	
198	00 08/23/105 11:10:53	1.231	0.864	1.271	
199	00 08/23/105 11:12:02	1.420	1.249	1.466	
200	00 08/23/105 11:29:58	0.284	0.096	0.293	
MAX:		5.775	4.706	4.203	
MIN:		0.284	0.096	0.293	
MEAN:		1.494	1.489	1.536	
SDEV:		0.537	0.538	0.669	

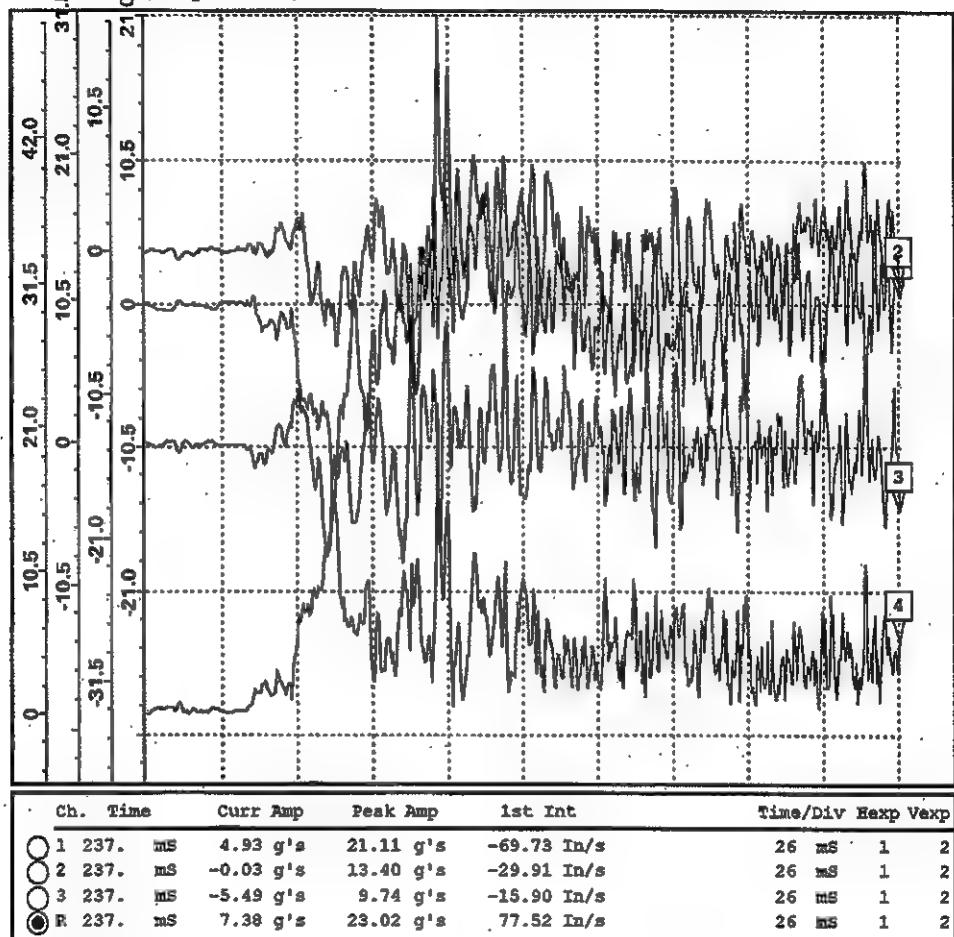
# GRAPH 1

## B-52 RADOME

### ROTATIONAL DROP TEST

Aug 18 2005 13:51      Test Engineer : Evans  
 Test type : Edgewise Drop      Impact Point : Forward edge  
 Container/Item: Aluminum/radome      Drop Height : 12 inches

V. Angle: 48.98; H.Angle: 269.69; Filter: = 200 Hz



PEAK G RESULTANT: 23 g's. PEAK G(X): 21 g's. 200Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vart.); Ch2 = Y(long.); Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature humidity.

ASTM D 4169, ASTM D 6179. SAE ARP1967.

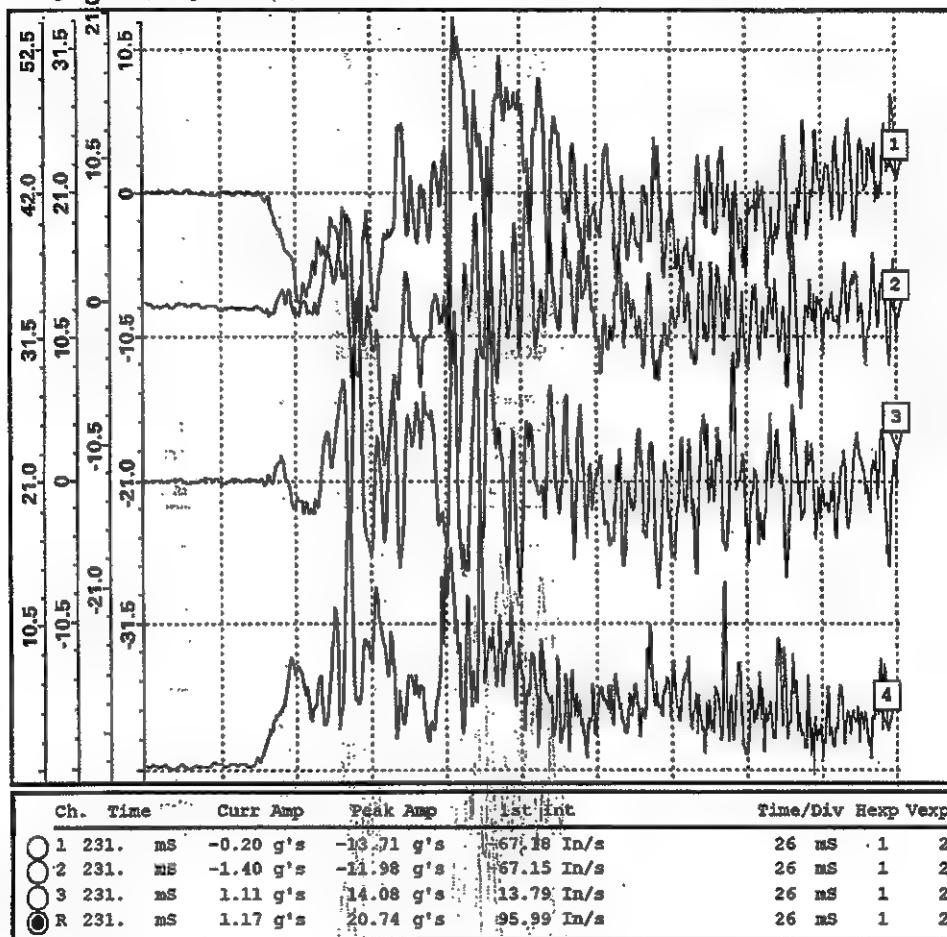
## GRAPH 2

### B-52 RADOME

#### ROTATIONAL DROP TEST

Aug 18 2005 13:57      Test Engineer : Evans  
 Test type : Cornerwise Drop      Impact Point : Forward-left corner  
 Container/Item: Aluminum/radome      Drop Height : 12 inches

V. Angle: 96.32; H.Angle: 141.71; Filter: = 200 Hz



PEAK g RESULTANT: 21 Gs. PEAK G(Z): 14Gs. 200Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vert.); Ch2 = Y(long.); Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature humidity.

ASTM D 4169, ASTM D 6179. SAE ARP1967.

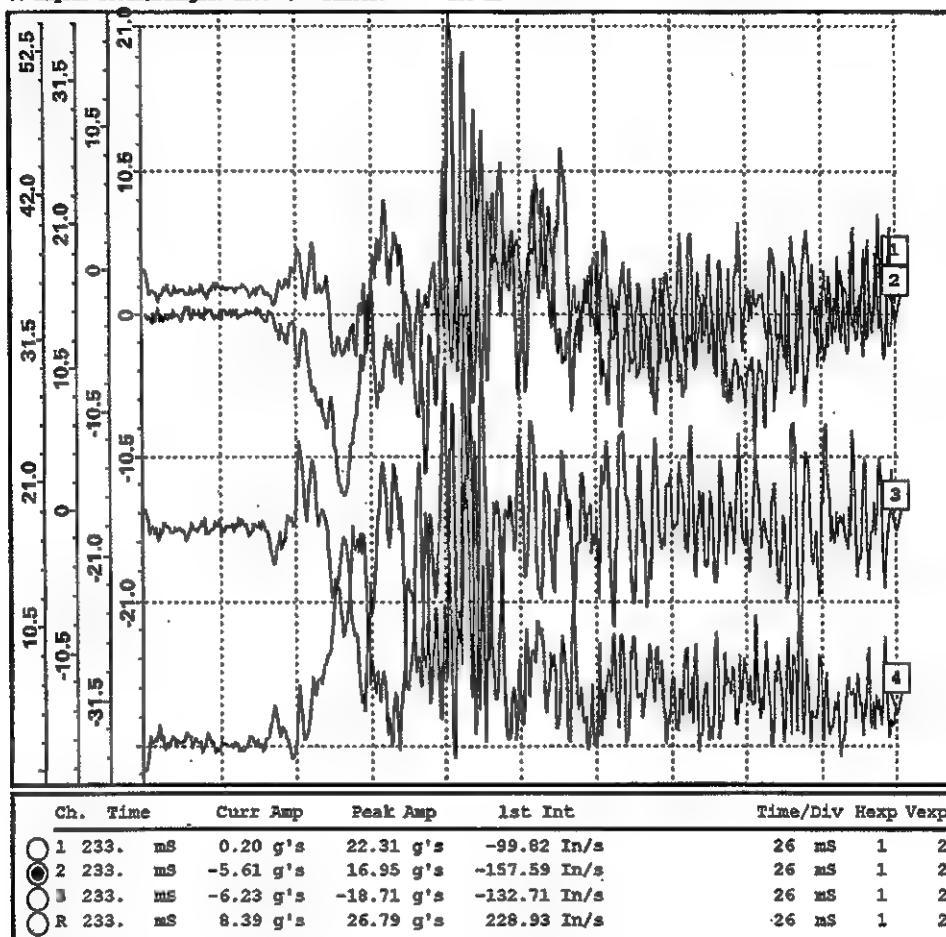
### GRAPH 3

## B-52 RADOME

### ROTATIONAL DROP TEST

Aug 18 2005 14:00      Test Engineer : Evans  
 Test type : Cornerwise Drop      Impact Point : Forward-right corner  
 Container/Item: Aluminum/radome      Drop Height : 12 inches

V. Angle: 98.62; B. Angle: 228.00; Filter: - 200 Hz



PEAK G RESULTANT: 27 Gs. PEAK G(X): 23 Gs. 200 Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vert.); Ch2 = Y(long.); Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature \_humidity.

ASTM D 4169, ASTM D 6179. SAE ARP1967.

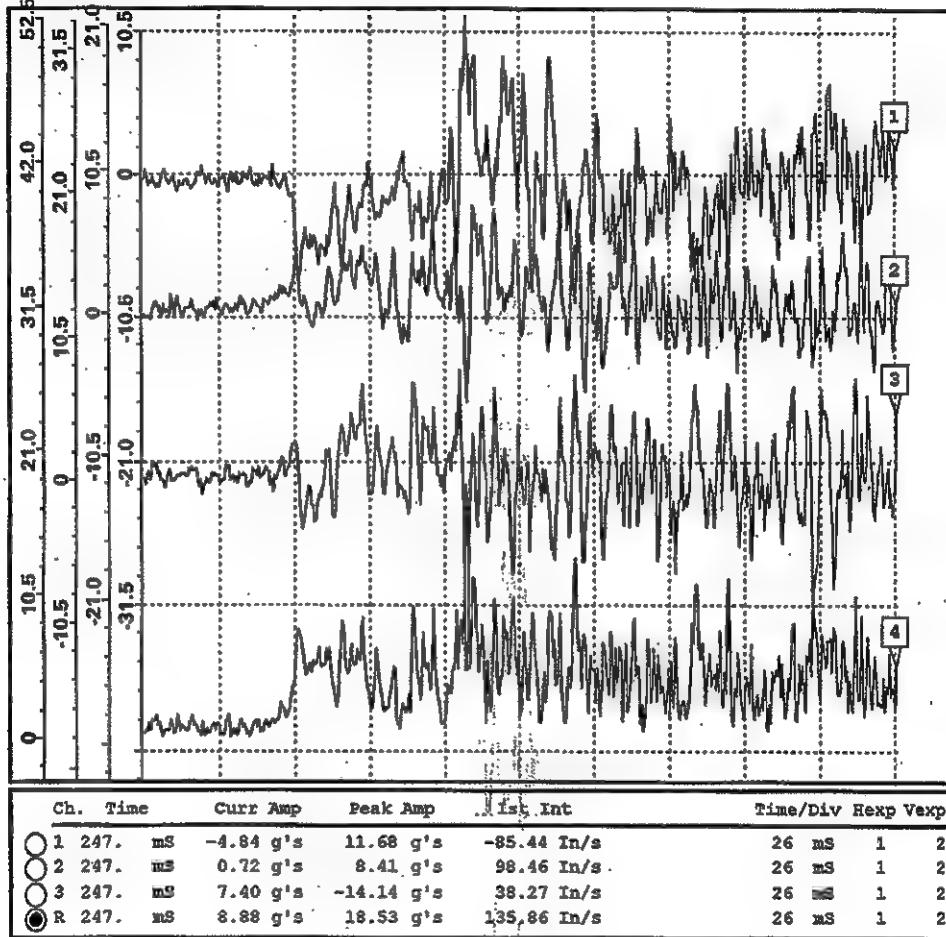
## GRAPH 4

### **B-52 RADOME**

#### ROTATIONAL DROP TEST

Aug 18 2005 14:04      Test Engineer : Evans  
 Test type : Edgewise Drop      Impact Point : Aft edge  
 Container/Item: Aluminum/radome      Drop Height : 12 inches

V. Angle: 123.05; H.Angle: 84.42; Filter: = 200 Hz



PEAK G RESULTANT: 19 Gs. PEAK G(Z): 14 Gs. 200 Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vert.); Ch2 = Y(long.); Ch3 = Z(trans.)  
Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature humidity.

ASTM D 4169, ASTM D 6179. SAE ARP1967.

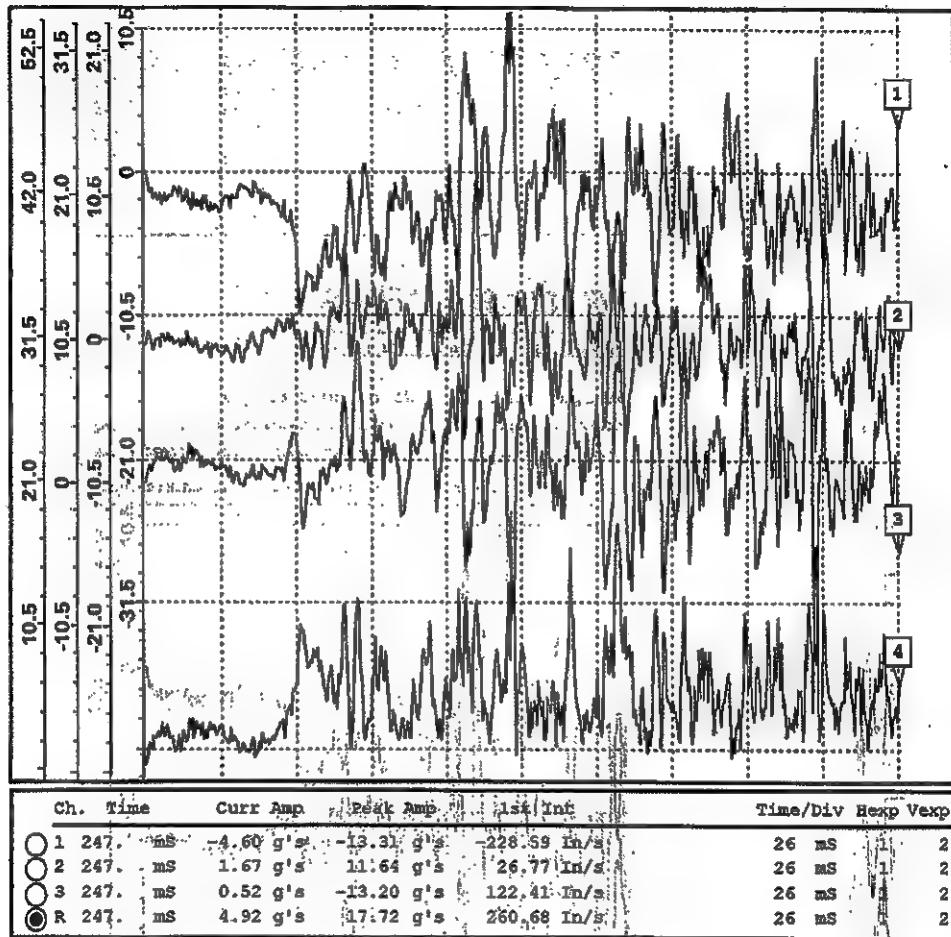
## GRAPH 5

### B-52 RADOME

#### ROTATIONAL DROP TEST

Aug 18 2005 14:12      Test Engineer : Evans  
 Test type : Cornerwise Drop      Impact Point : Aft-left corner  
 Container/Item: Aluminum/radome      Drop Height : 12 inches

V. Angle: 159.12; H. Angle: 17.27; Filter: = 200 Hz



PEAK G RESULTANT: 18 Gs. PEAK G(Z) : 13 Gs. 200 Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vert.); Ch2 = Y(long.); Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature humidity.

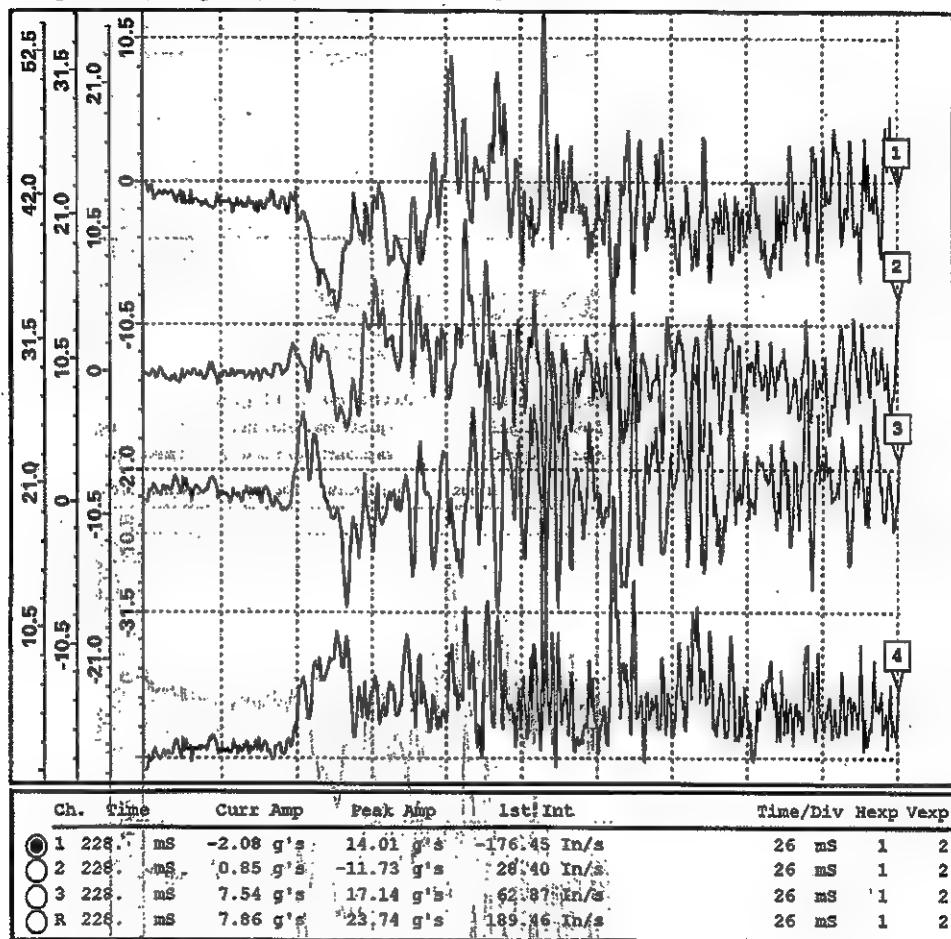
ASTM D 4169, ASTM D 6179, SAE ARP1967.

## GRAPH 6

### B-52 RADOME

#### ROTATIONAL DROP TEST

Aug 18 2005 14:10      Test Engineer : Evans  
 Test type : Cornerwise Drop      Impact Point : Aft-right corner  
 Container/Item: Aluminum/radome      Drop Height : 12 inches  
 V. Angle: 105.33; H.Angle: 83.60; Filter: = 200 Hz



PEAK G RESULTANT: 24 g's, PEAK G(Z): 17 g's, 200 Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vert.); Ch2 = Y(long.); Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature \_humidity.

ASTM D 4169, ASTM D 6179, SAE ARP1967

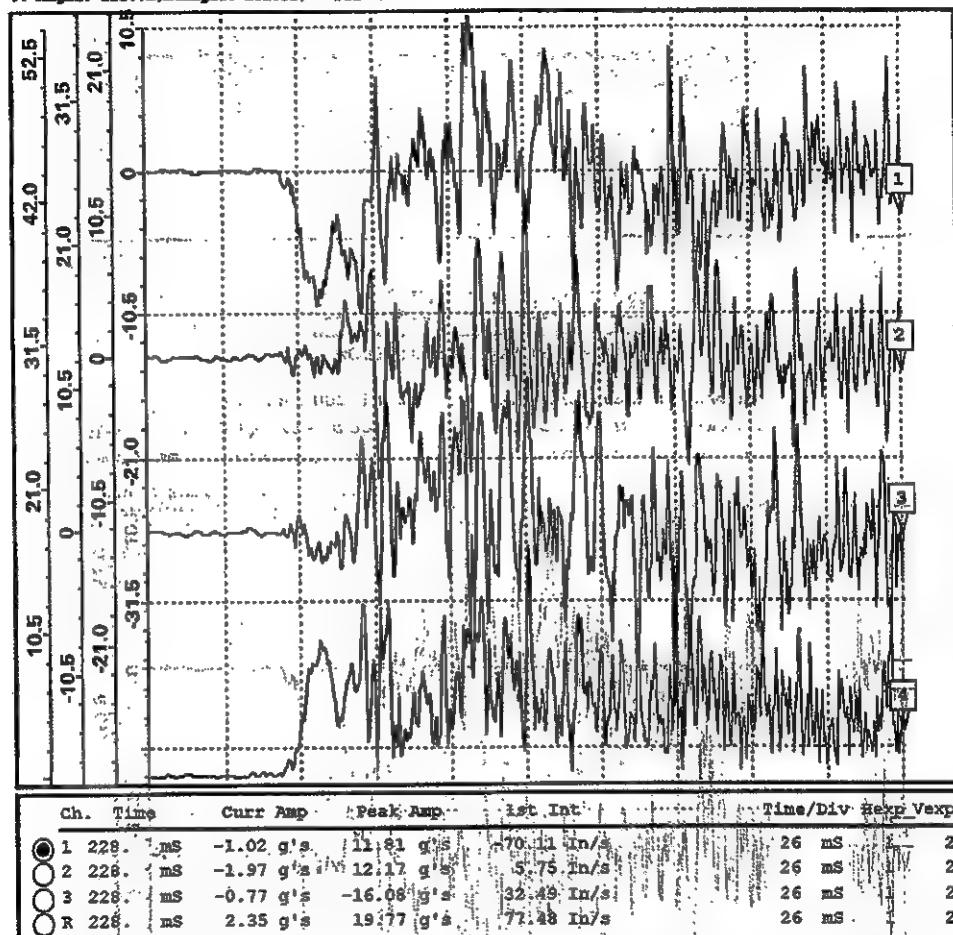
## GRAPH 7

### B-52 RADOME

#### ROTATIONAL DROP TEST

Test type : Edgewise Drop      Aug 18 2005 14:37      Test Engineer : Evans  
 Container/Item: Aluminum/radome      Impact Point : Left edge  
 Drop Height : 12 inches

V. Angle: 115.72; H.Angle: 201.32; Filter: = 200 Hz



PEAK G RESULTANT: 20 Gs. PEAK G(2): 16 Gs. 200 Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vart.) ; Ch2 = Y(long.) ; Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature \_humidity.

ASTM D 4169, ASTM D 6179. SAE ARP1967.

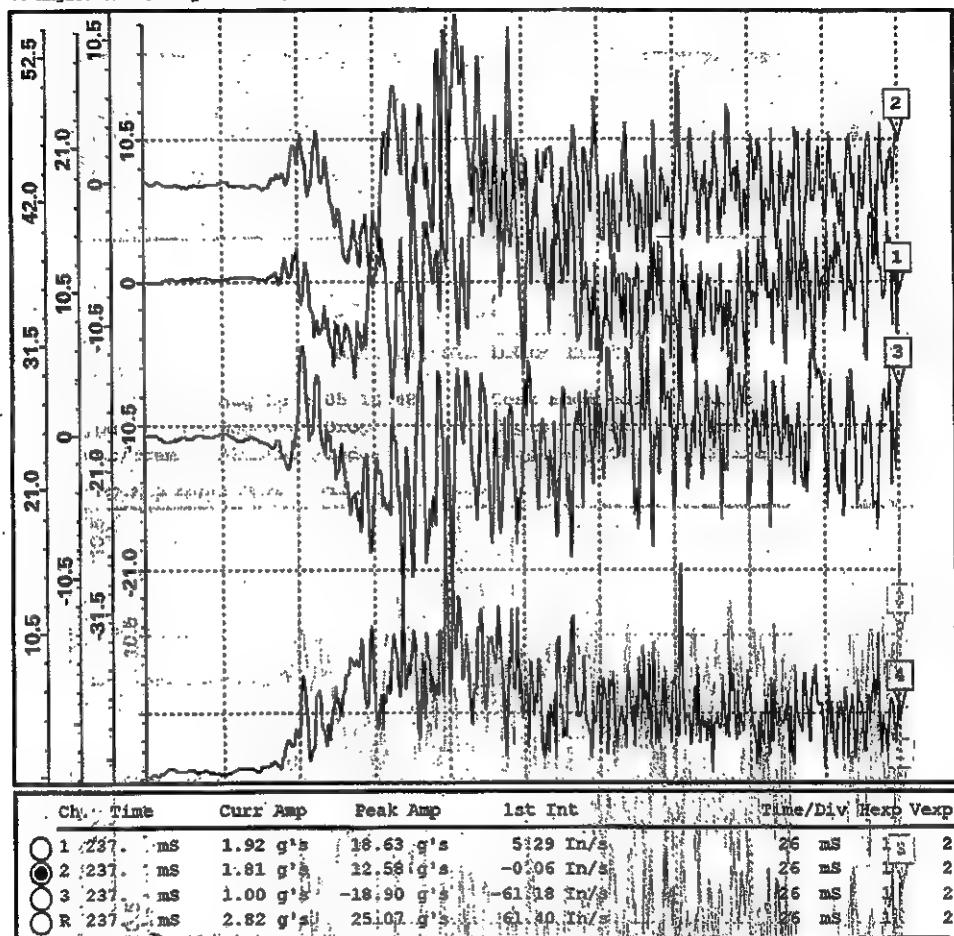
## GRAPH 8

### B-52 RADOME

#### ROTATIONAL DROP TEST

Aug 18 2005 14:48      Test Engineer : Evans  
 Test type : Edgewise Drop      Impact Point : Right edge  
 Container/Item: Aluminum/radome      Drop Height : 12 inches

V. Angle: 47.23; H.Angle: 28.35; Filter: = 200 Hz



PEAK G RESULTANT: 26 g's. PEAK G(Z/X): 19 g's. 200 Hz filter.

ACCELEROMETER OUTPUT: Ch1 = X(vert.); Ch2 = Y(long.); Ch3 = Z(trans.)

Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature humidity.

ASTM D 4169, ASTM D 6179. SAE ARP1967.

## GRAPH 9

### B52 RADOME

#### PENDULUM IMPACT TEST

Jan 4 2006 12:44

Test Engineer : Evans

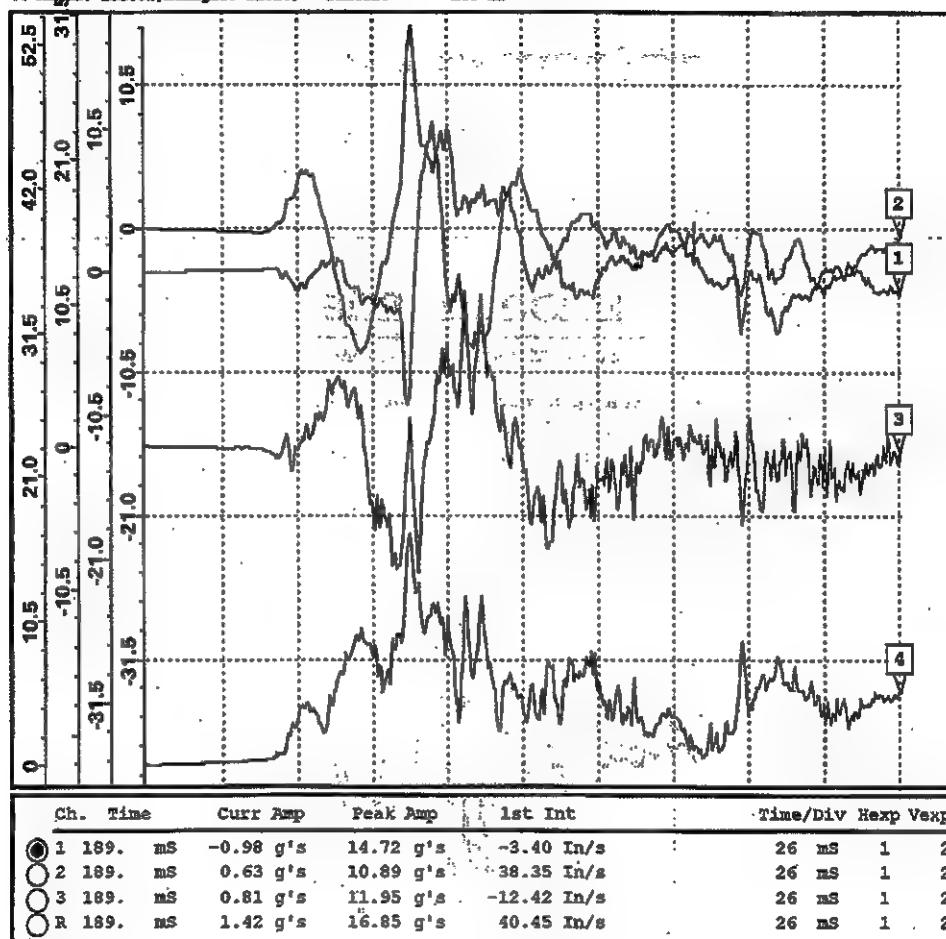
Test type : Impact

Impact Point : Forward side

Container/Item: Aluminum/B52 RADOME

Impact Velocity: 7.3 ft/s

V. Angle: 133.71; H.Angle: 52.16; Filter: = 200 Hz



PEAK G RESULTANT: 17 G's. PEAK G(X): 15 G's.

ACCELEROMETER OUTPUT: Ch1 = X(long.) ; Ch2 = Y(vert.) ; Ch3 = Z(trans.)  
Ch4 = Resultant.

Aft side = desiccant port end.

Ambient temperature humidity.

ASTM D 4169, ASTM D 5179. SAE ARP1967. Accel S/N 16471

**APPENDIX 3**  
**FIGURES**

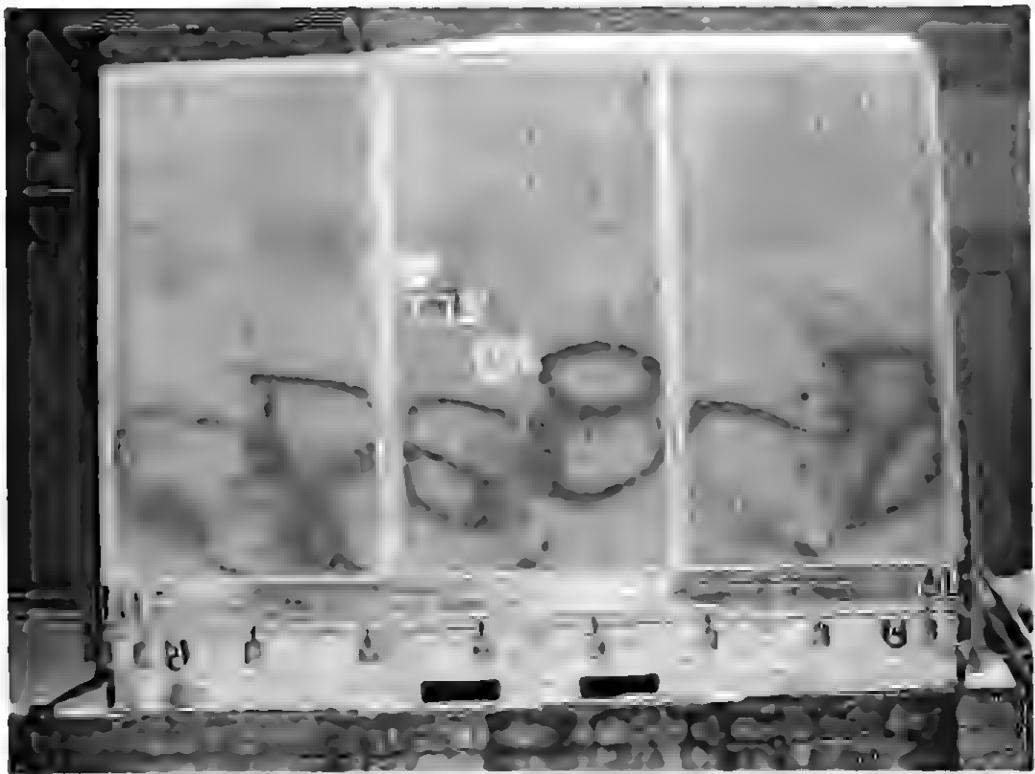


Figure 1. B-52 Nose Radome Container (Side View)



Figure 2. B-52 Nose Radome Container (End View)

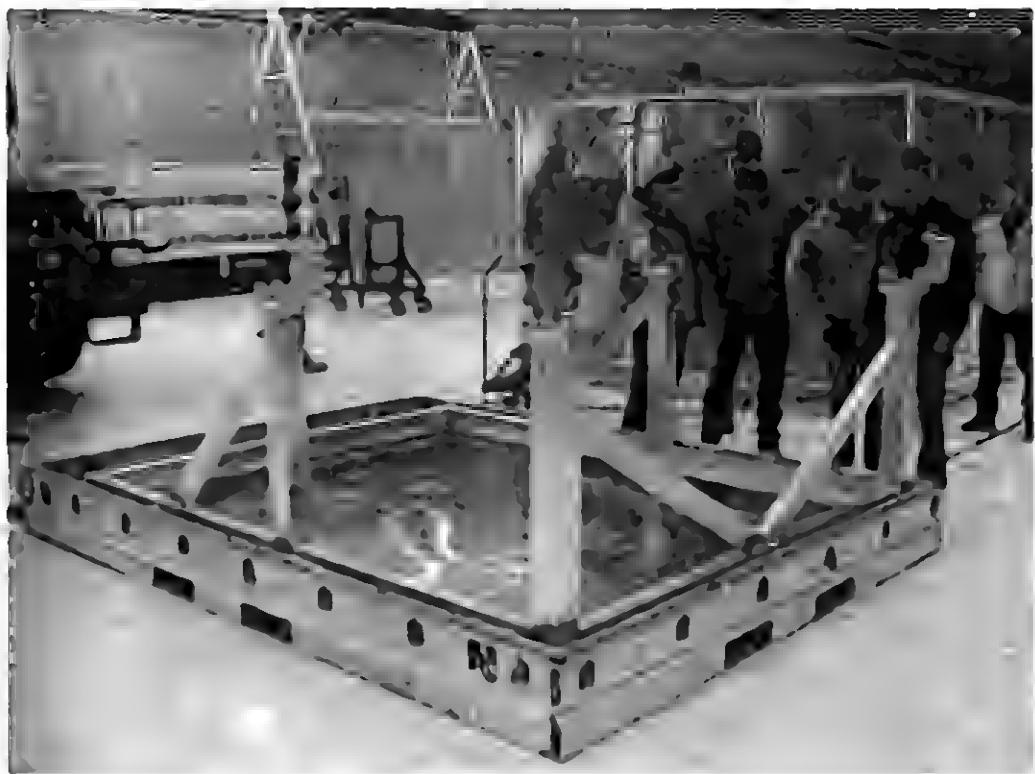


Figure 3. Container Interior



Figure 4. B-52 Nose Radome with Lifting Frame attached in transport trailer

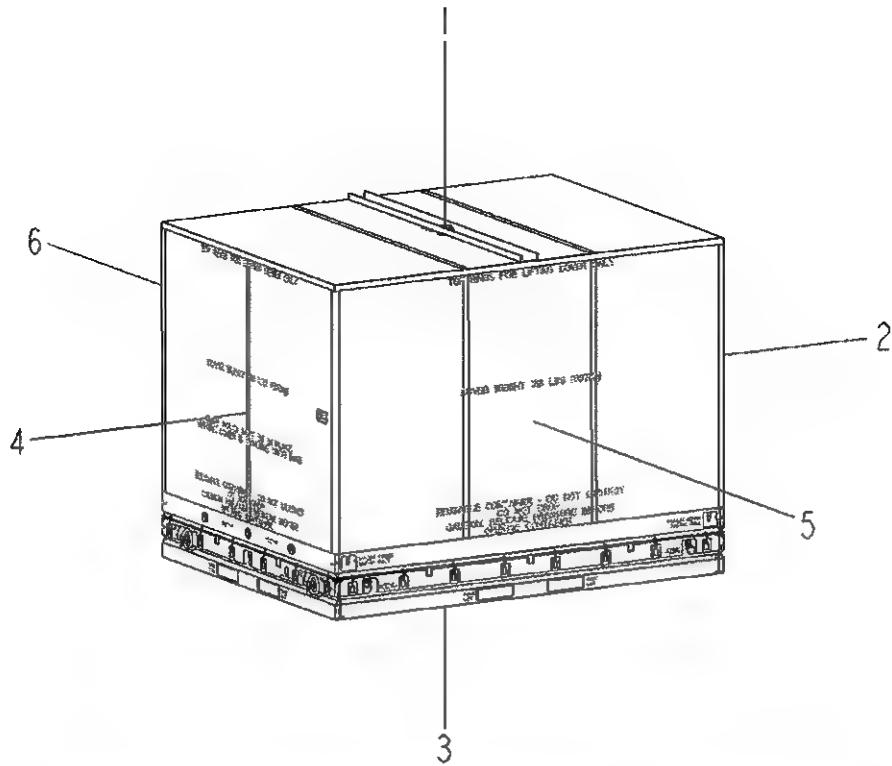


Figure 5. Container Side Designations

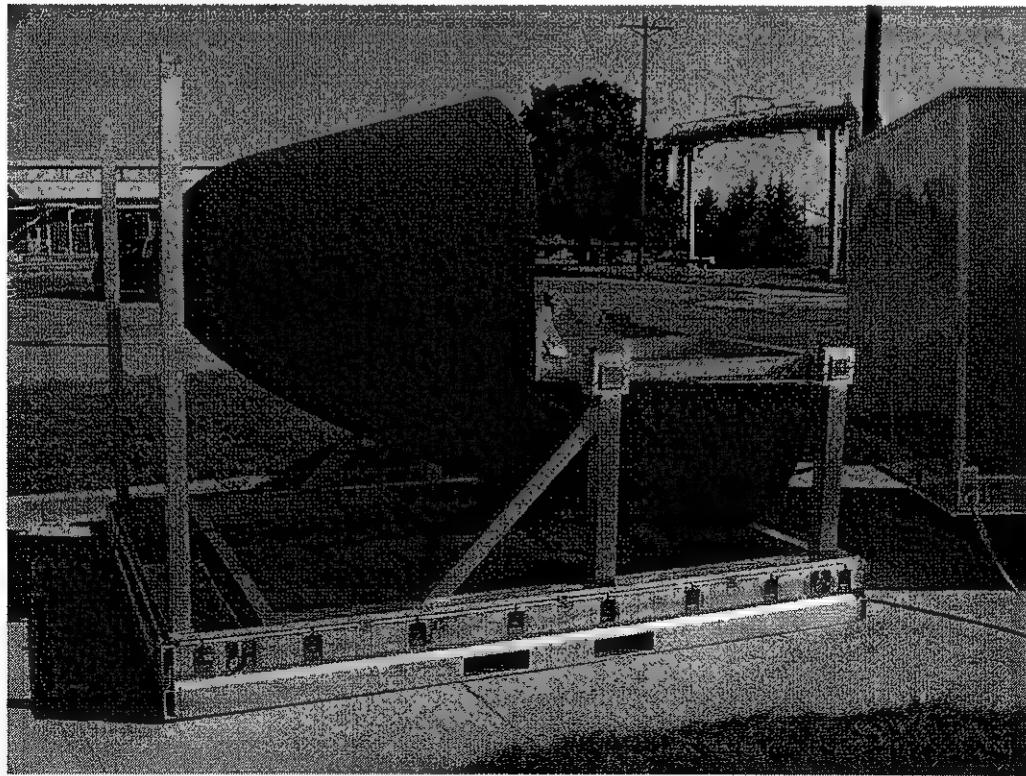


Figure 6. B-52 Nose Radome in Container Base



Figure 7. Container Weight

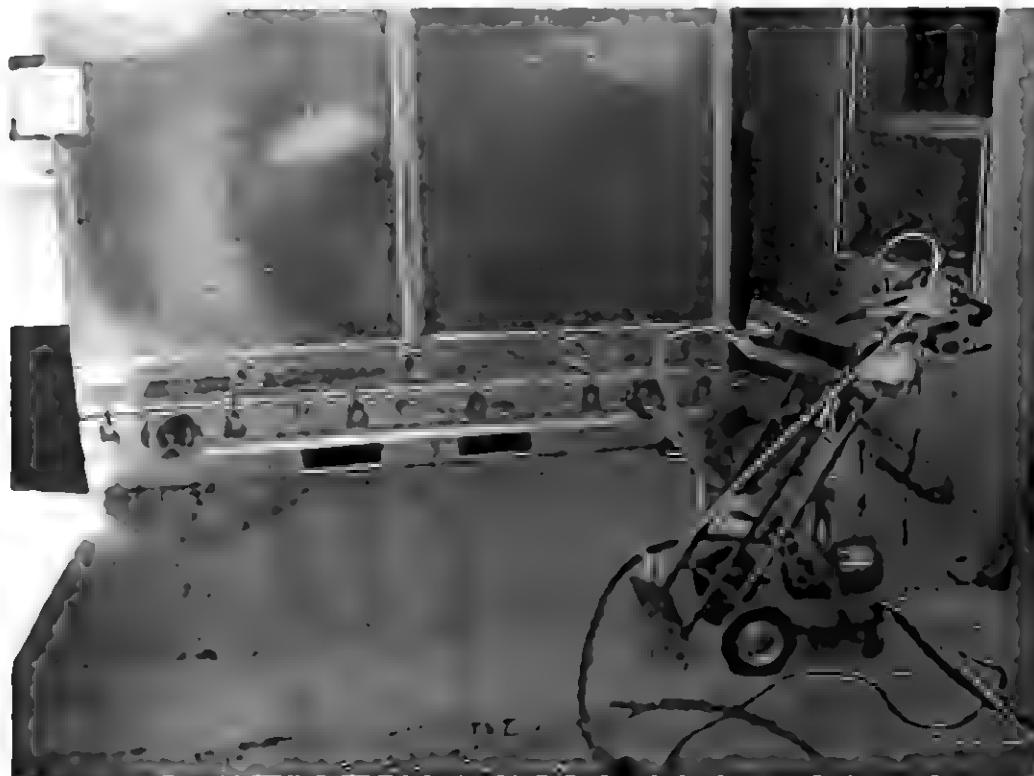


Figure 8. Pneumatic Pressure Leak Test



Figure 9. Cornerwise-Drop Test



Figure 10. Edgewise-Drop Test



Figure 11. Pendulum-Impact Test



Figure 12. Over-the-Road Vehicle Vibration Test

**APPENDIX 4**

**DISTRIBUTION LIST**

DISTRIBUTION LIST

DTIC/O  
DEFENSE TECHNICAL INFORMATION CENTER  
FORT BELVOIR VA 22060-6218

HQ AFMC/LSO/LO  
WRIGHT-PATTERSON AFB OH 45433-5540

OC-ALC/GBMSTP  
TINKER AFB OK 73145-5000

OO-ALC/LGMPD  
HILL AFB UT 84056-5000

WR-ALC/LGMTP  
ROBINS AFB GA 31098-5000

OC-ALC/MNBDA  
TINKER AFB OK 73145-5000

HQ ACC/A4A52  
TINKER AFB OK 73145-3021

327 BMSG/LR  
TINKER AFB OK 73145-3021

**APPENDIX 5**  
**REPORT DOCUMENTATION**

REPORT DOCUMENTATION PAGE					Form Approved OMB No. 0704-0188	
<p>The public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing the burden, to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number.</p> <p><b>PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.</b></p>						
1. REPORT DATE (DD-MM-YYYY)		2. REPORT TYPE		3. DATES COVERED (From - To)		
30-01-06		Final		Sep 2004 - Jan 2006		
4. TITLE AND SUBTITLE					5a. CONTRACT NUMBER	
Development of the B-52 Nose Radome Container, CNU-680/E					5b. GRANT NUMBER	
					5c. PROGRAM ELEMENT NUMBER	
6. AUTHOR(S)					5d. PROJECT NUMBER	
Joel A. Sullivan Susan J. Evans					04-P-111	
					5e. TASK NUMBER	
					5f. WORK UNIT NUMBER	
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)					8. PERFORMING ORGANIZATION REPORT NUMBER	
HQ AFMC/LSO/LOP 5215 Thurlow St. Wright-Patterson AFB, OH 45433-5440					06-R-01	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)					10. SPONSOR/MONITOR'S ACRONYM(S)	
327 BMSG/LR 3001 Staff Dr. Suite, 2AG 192B Tinker AFB, OK 73145-3021					11. SPONSOR/MONITOR'S REPORT NUMBER(S)	
12. DISTRIBUTION/AVAILABILITY STATEMENT						
Approved for public release Distribution unlimited						
13. SUPPLEMENTARY NOTES						
14. ABSTRACT						
<p>This report is responsible for documenting the design and qualification testing of the CNU-680/E container. The container developed will protect the Nose Radome mechanically, environmentally, and make the item much more easy to maneuver during worldwide shipment and storage. The CNU-680/E, designed per ARP1967A, is an aluminum, long life, controlled breathing, reusable container. The container passed all qualification tests ASTM D4169 as well as field tests. The CNU-680/E container will not only meet the users' requirements but will also provide an economic savings in O&amp;M costs. The CNU-680/E container was designed, prototyped and tested in house at the Air Force Packaging Technology &amp; Engineering Facility and is qualified for production release.</p>						
15. SUBJECT TERMS						
CNU-680/E, B-52 Nose Radome, Aluminum Container, Reusable Container, Long-Life Container						
16. SECURITY CLASSIFICATION OF:			17. LIMITATION OF ABSTRACT		18. NUMBER OF PAGES	
a. REPORT b. ABSTRACT c. THIS PAGE			UU		45	
U U U			UU		45	
					19a. NAME OF RESPONSIBLE PERSON Joel A. Sullivan	
					19b. TELEPHONE NUMBER (Include area code) (937) 257-8162	